



# Session Descriptions

## May 16, 2001--Day 1

### Pioneer Technology Innovation:

In order to develop the aerospace systems of the future, revolutionary approaches to system design and technology development will be necessary. Pursuing technology fields that are in their infancy today, developing the knowledge bases necessary to design radically new aerospace systems, and performing efficient, high-confidence design and development of revolutionary vehicles are challenges that face us in innovation. These challenges are intensified by the demand for safety in our highly complex aerospace systems. The goal to Pioneer Technology Innovation is unique in that it focuses on broad, crosscutting innovations critical to a number of NASA missions and to the aerospace industry in general. NASA's latest accomplishments include demonstration of a heterogeneous distributed computing environment, a test of a shape memory alloy to control flow into an engine inlet, and a full-scale test of a composite wing box that performed as predicted.

**Moderator:** Dr. William Ballhaus, Jr., President, The Aerospace Corp.

### Panel Members:

Mr. Howard Bloom, NIST Director of Manufacturing Engineering Laboratory (MEL)

Dr. Rich Wlezien, Program Manager, Quiet Supersonic Transport (QST), DARPA

Dr. Richard E. Smalley, Director, Rice Center for Nanoscale Science & Technology, Rice University

Mr. Michael Hudson, Vice Chairman, Rolls-Royce North America

Dr. Darrel Tenney, Director, Aerospace Vehicle Systems, NASA-LARC



## **NASA Turning Goals Into Reality Conference Innovation in Aerospace Transportation**

# **Pioneering Technology Innovation in Aerospace Vehicle Systems**

Dr. Darrel R. Tenney  
Director Aerospace Research Office  
NASA Langley Research Center

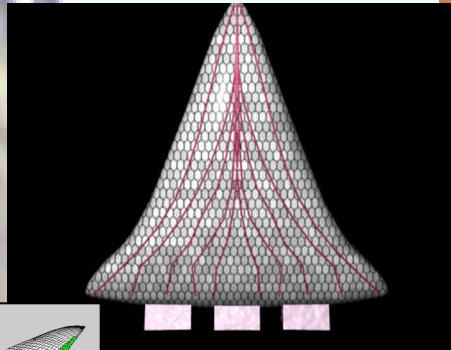
Ronald Reagan International Trade Center  
Washington, D.C.  
May 15-17, 2001

# Towards Advanced Aerospace Vehicles

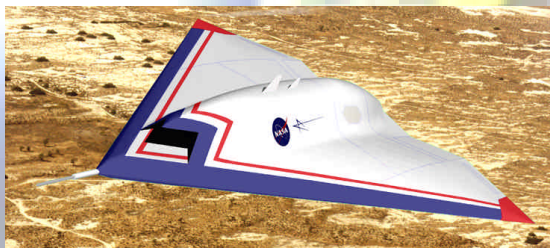
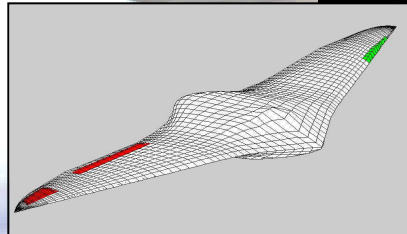
Aerospace Vehicle Includes:

- Distributed self-assessment and repair
- Real-time, multi-point reconfiguration
- Adaptive virtual (aerodynamic) shape change
- Adaptive structural shape change
- Biologically-inspired concepts
- Seamless wings
- Multi-functional, active structures

Advanced Technology Development



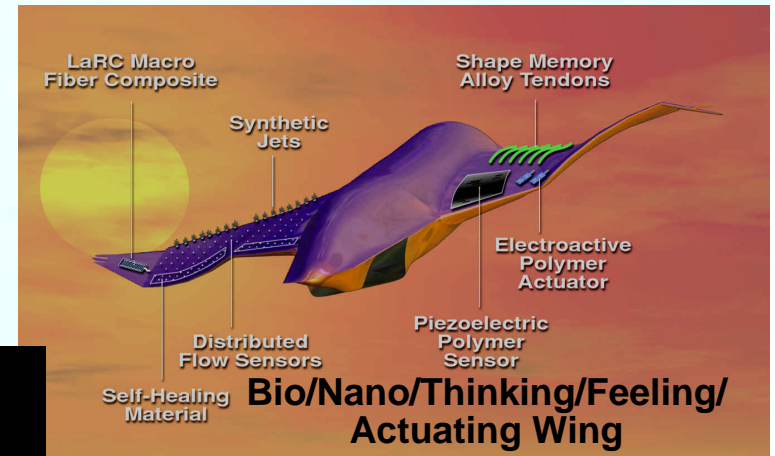
Smart Composite wing with Active Flow Control



Modern Advanced Airfoil Metal Wing



Time



Smart Flow Control Wing with Holistic Health Monitoring

Aeroelastically Tailored Wet Composite Wing





**Efficiency:  
Wingtip Flow  
Control**

**Safety:  
Full Situational  
Awareness**

**Propulsion:  
Highly Distributed  
and Ultra-Efficient**

**Survivability:  
Distributed Nervous System  
Self-Healing Systems**

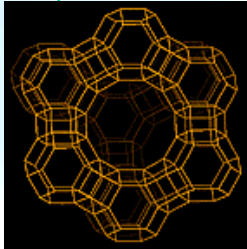
**Morphing:  
Continuous Optimal  
Shape Control**

**Strong, Lightweight:  
Integral Wing-Body  
Structure**

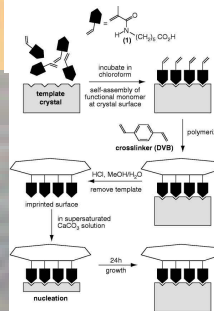
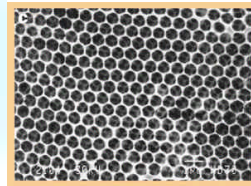
**Agility:  
Extreme  
Maneuverability**

# Revolutionary Technologies

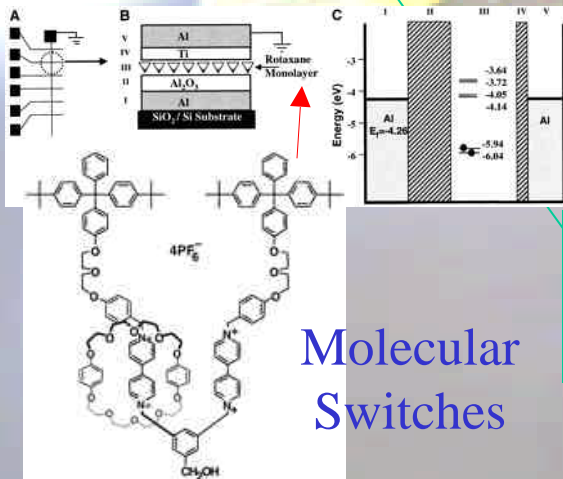
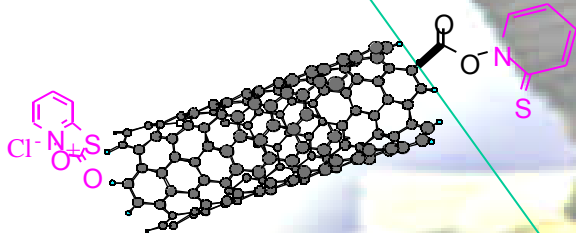
## Biotechnology



### Templates



### Functionalization

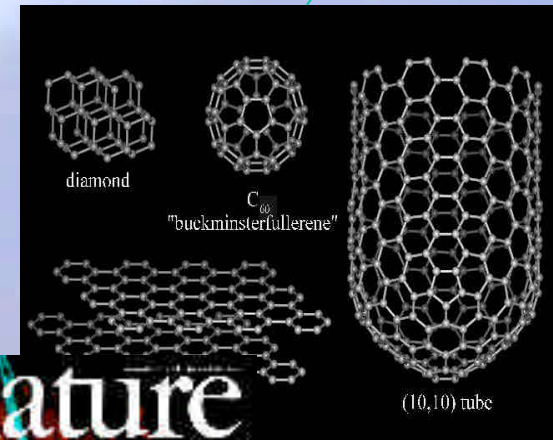


### Molecular Switches

## Information Technology

## Nanotechnology

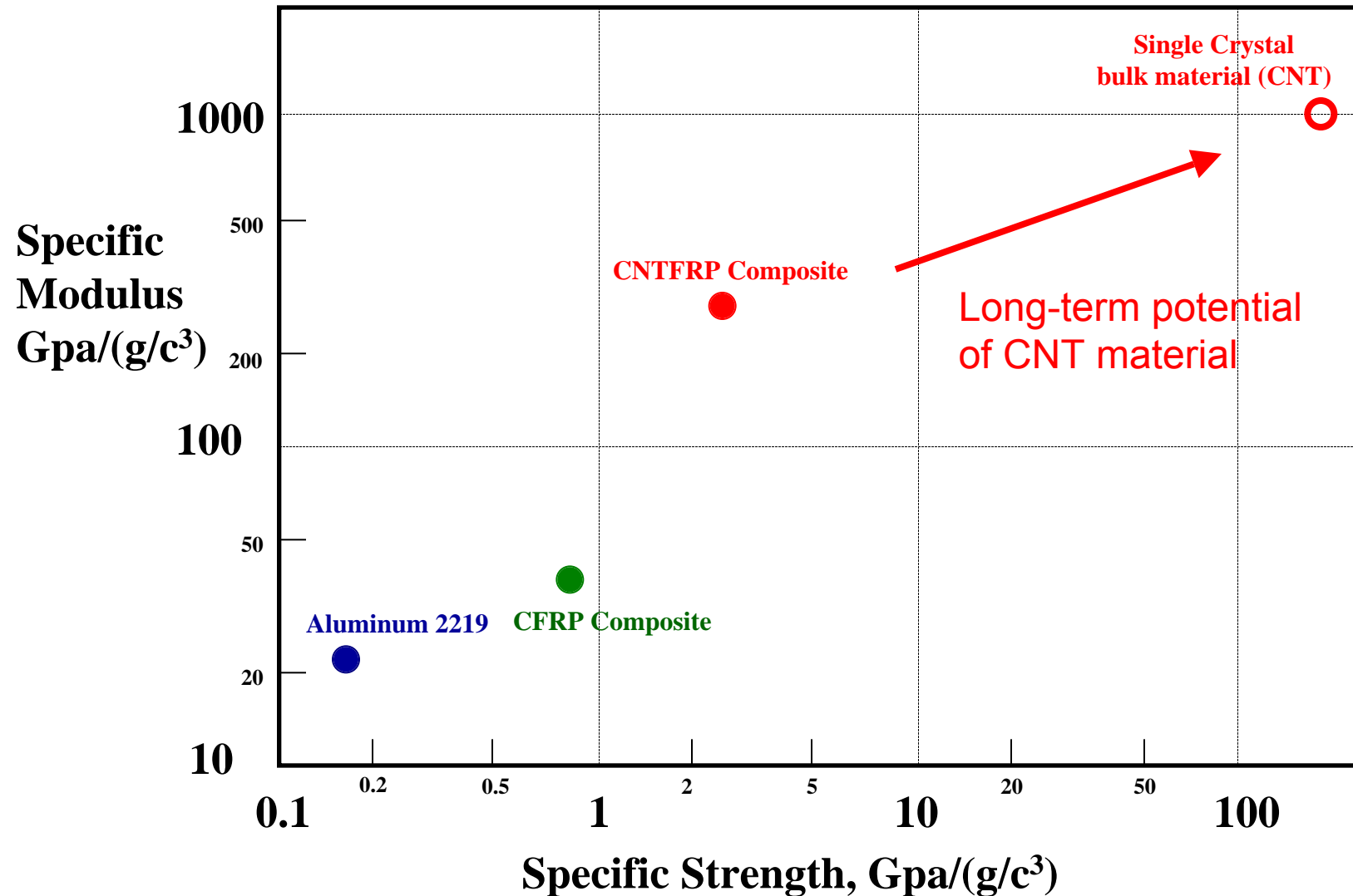
### The Forms of Carbon



### Nanotubes

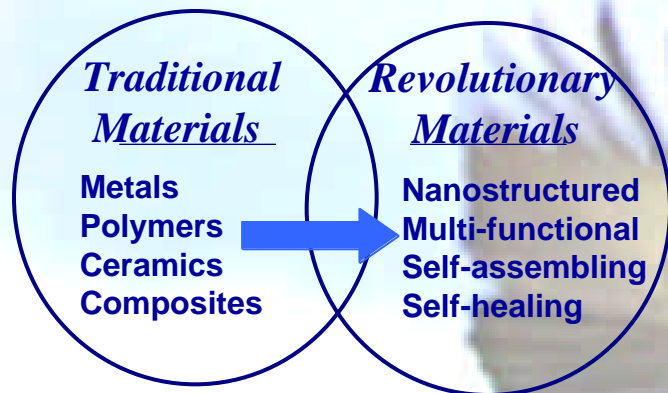
# Properties of Carbon Nanotubes (CNT)

● Baseline Material, available today    ● Best available, under development    ● emerging material, carbon nanotubes

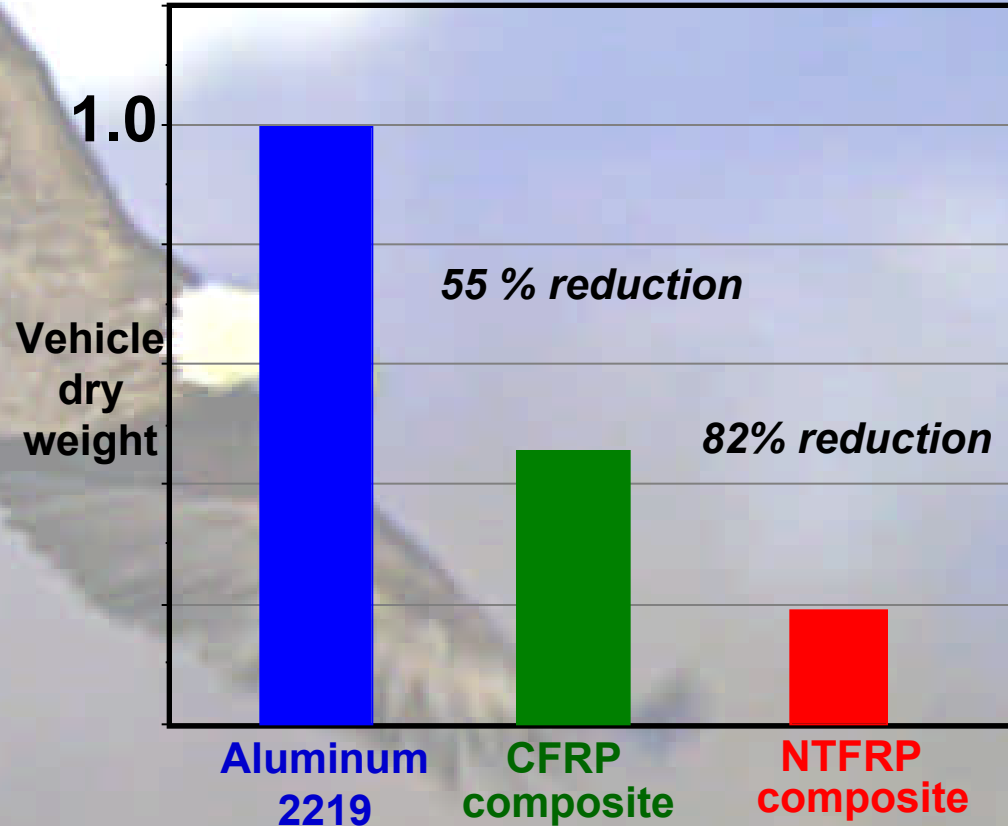




# Carbon Nanotube Composites for Aerospace Vehicles



The Nanotube Fiber Reinforced Polymer Composites (NTFRP) compared to an advanced Carbon Fiber Reinforced Polymer (CFRP) composite

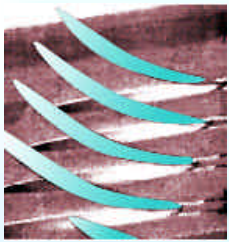


**Reusable Launch Vehicles**



# A Wing for All Flight Regimes

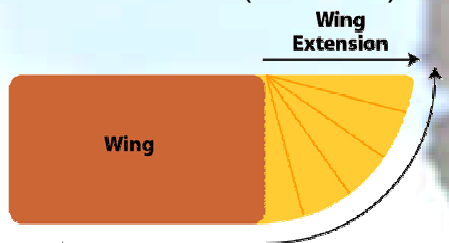
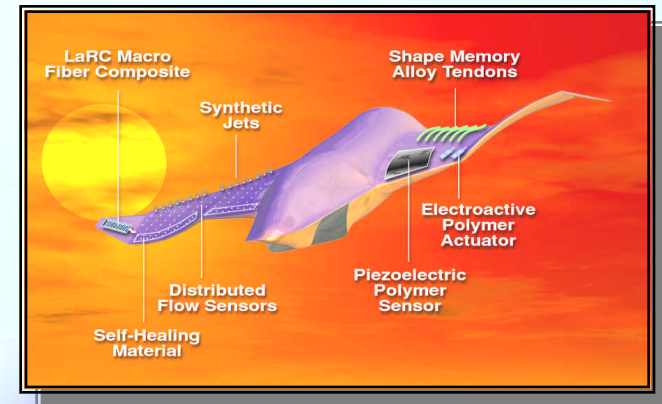
## One for all and All by One



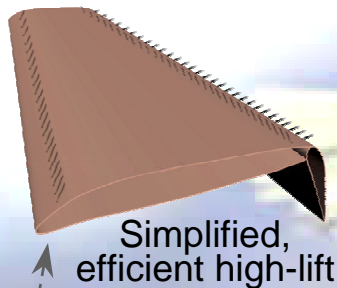
Maneuvering with micro-surface actuators (feathers)

### Through advanced technologies:

- Compliant surfaces
- Plasma actuators
- Synthetic jets
- Fluidic thrust vectoring
- Active materials
- MEMs control surfaces
- Adaptive structures



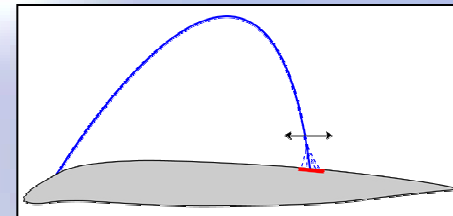
Variable wing span



Simplified, efficient high-lift



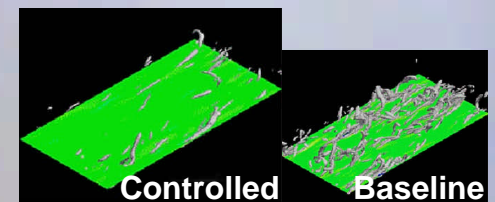
Camber change



Shock control with active compliant skin



LE radius and airfoil thickness variability



Controlled

Baseline

Turbulence control with adaptive wing skin



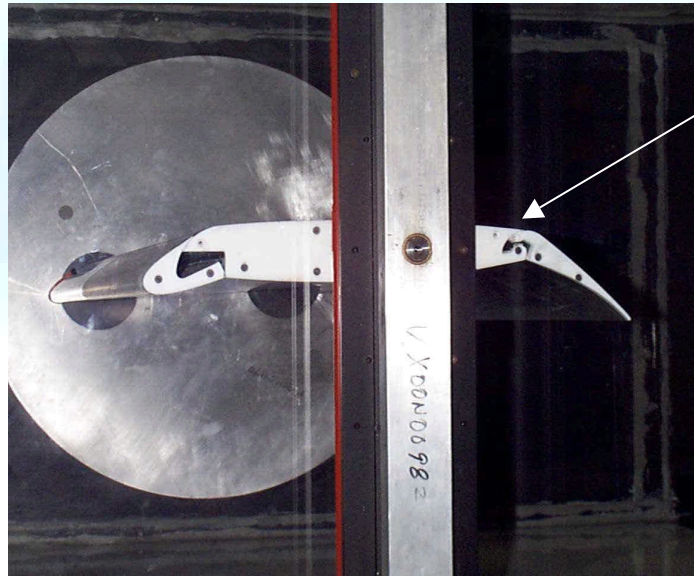
Today's Wing

Enhanced control with wing twist

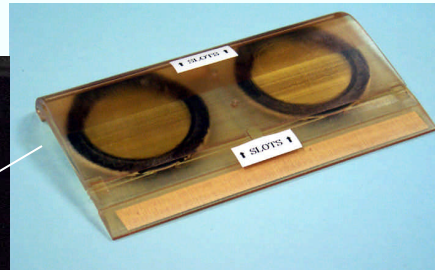
Efficient, hingeless control surfaces



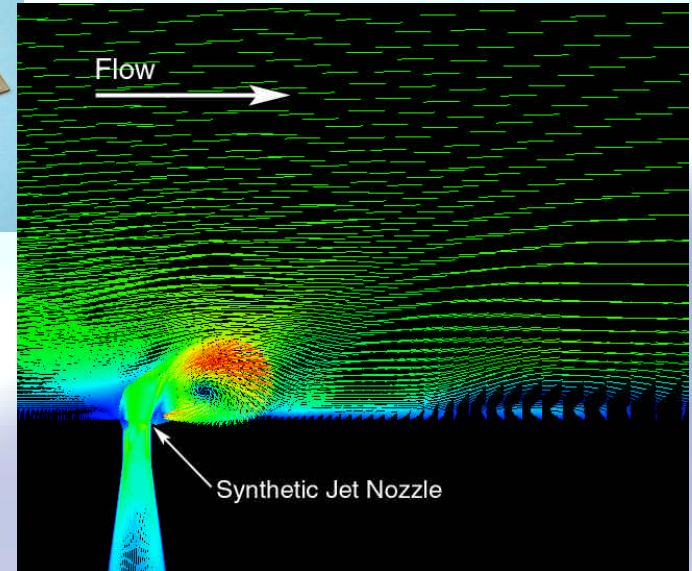
# Sample of NASA Langley Research in Active Flow



Model installed in tunnel



Synthetic Jet



Numerical simulation of synthetic jet

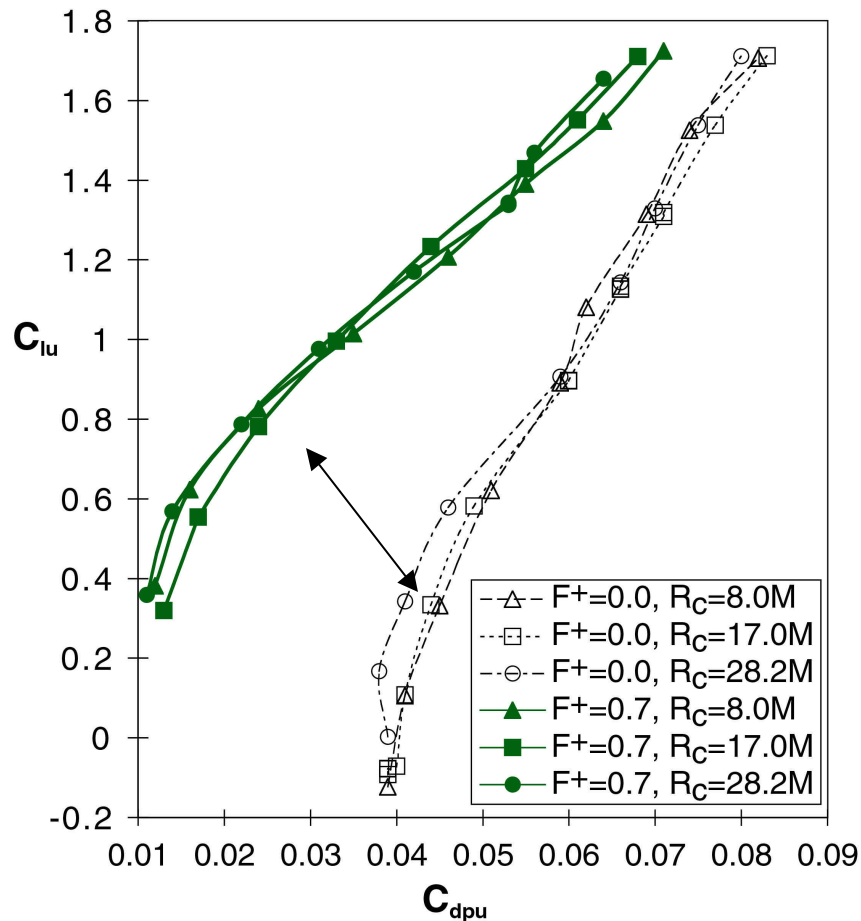
## Simplified High Lift System

- Integrates many of the advanced technology development areas at Langley
- Langley has demonstrated separation control to actual flight conditions
- Periodic excitation produces benefits with 100X less energy input
- Increased lift by 20-70%, decreased drag by 30-50%, improved stall characteristics

## Synthetic Jet Actuators

- Electrically driven by vibrating membranes in cavity using smart materials
- Adds unsteady momentum and vorticity to external flow through jet nozzle
- Fast actuation for excellent control response
- Synthetic jets provide additional weight benefit by requiring no plumbing or pumps
- Small and redundant for improved system reliability and safety

# Takeoff/Landing Applications: Separation Control



## Takeoff & Landing Conditions

Suction & Blowing Slot



- 30-50% Drag Reduction
- 20-70% Lift Increase
- 2 orders of magnitude less mass flow required compared to steady blowing
- Independent of Reynolds number

# NASA Simplified High Lift System: Motivation

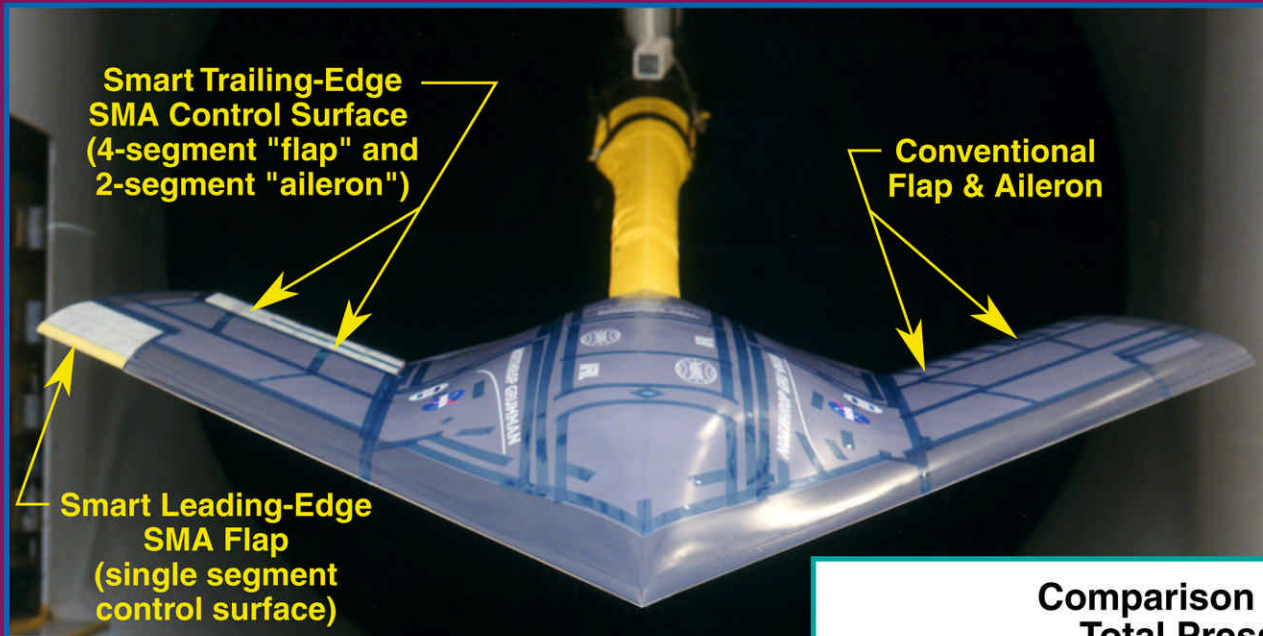
## (System Study Results)



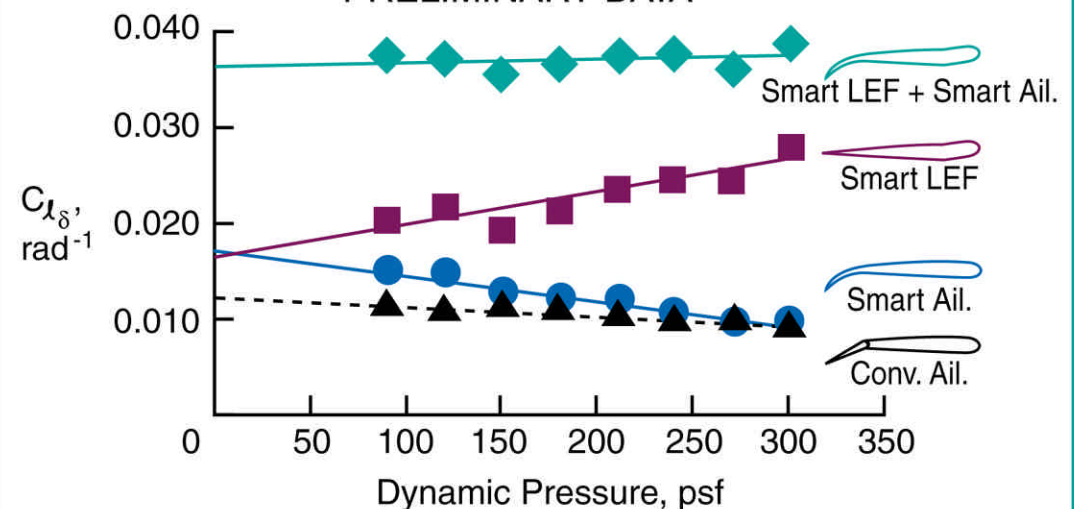
- **Two industry system studies confirm that simplified high lift systems provide the highest payoff for conventional vehicle designs**
- **\$410K manufacturing cost savings on \$30M airplane**
  - 2.6% reduction in part count
  - 3.3% reduction in empty weight
- **Yearly savings of \$45M in fuel costs per airline**
  - Based on estimates of Delta Airlines 1998 and 1999 fuel consumption
  - 3.3% reduction in cruise drag
- **More significant gains possible with vehicles designed from the conceptual stage with integrated active flow control**



# DARPA/AFRL/NASA/NORTHROP GRUMMAN SMART WING PHASE II, TEST 1 COMPLETED IN THE TDT



Comparison of Roll Effectiveness  
Total Pressure (H) = 1070 psf  
PRELIMINARY DATA



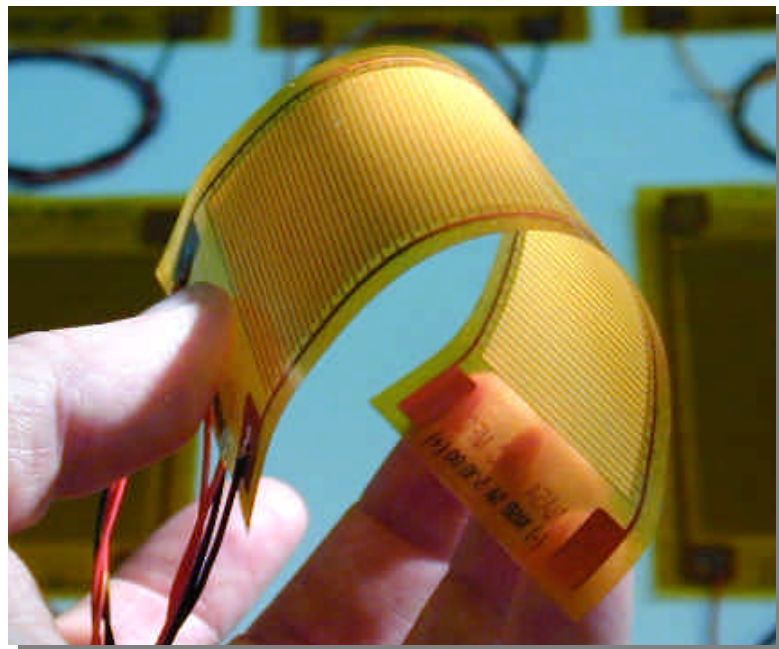


# LaRC Macro-Fiber Composite -FY00

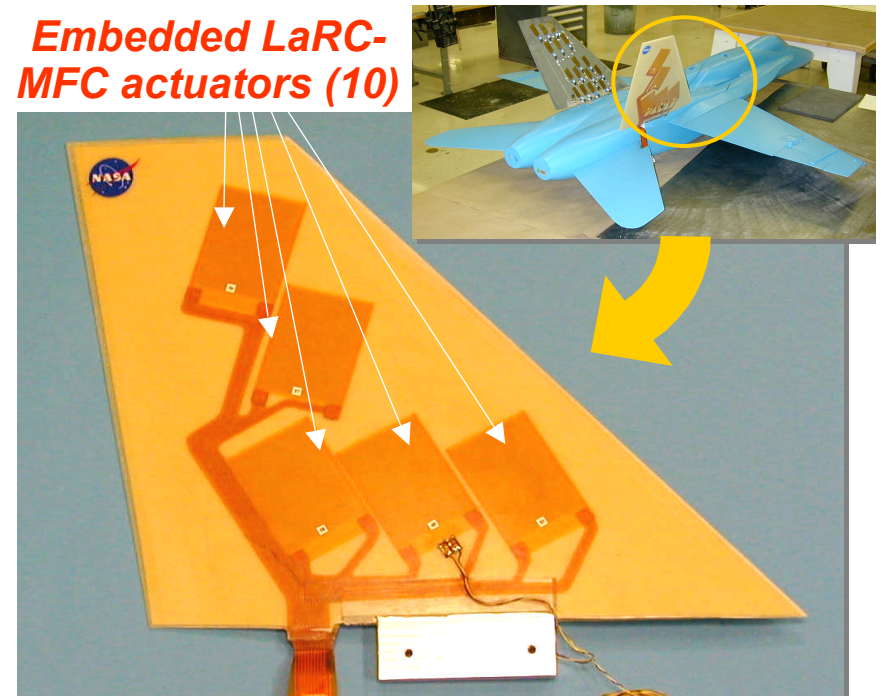


- R&D 100 Award and Editor's Choice
- Vibration Suppression Demonstrated in an Inflatable Strut
- Buffeting Alleviation Demonstrated in Wind-Tunnel Test (L2 Milestone)

**Z-folded strut containing LaRC-MFC actuators**



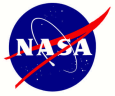
**Embedded LaRC-MFC actuators (10)**



**LaRC-MFC 1/6th scale active tail model**

# Macro-Fiber Composite

- 2000 R&D 100 Award and Editors' Choice
- Flexible, durable, cost-competitive, easily manufactured piezoelectric device
- Conforms to surfaces and can be embedded in or attached to flexible structures
- Senses strain and moves to dampen vibration or position smart structures



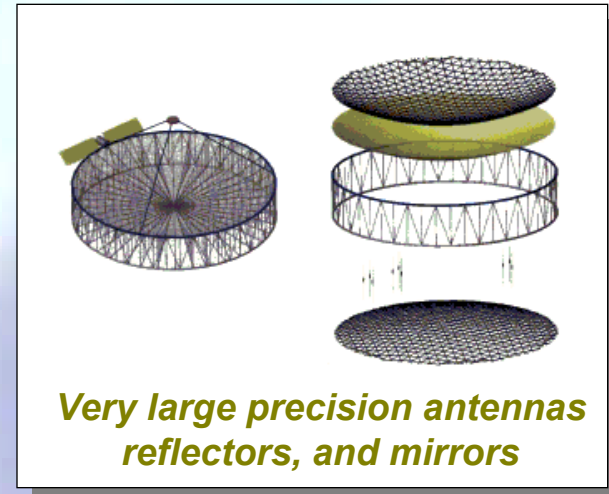
# Smart, Ultralight Deployable Spacecraft



**1. LaRC-MFC devices embedded in inflatable composite struts and folded for launch**

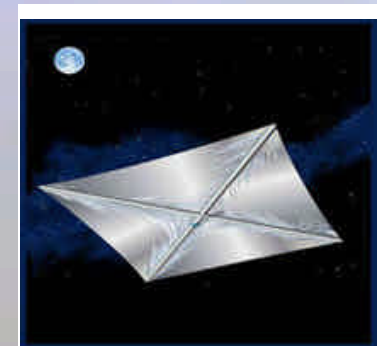


**2. Struts inflated and cured on-orbit**



**Very large precision antennas reflectors, and mirrors**

**3. Used as structure in ...**



**Solar sails with controllable spars**

***20 Fold Reduction in Vibration Using Smart Actuators  
Enables Deployable Structures Technology***



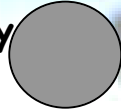
# Noise Attenuation

J. Posey

## Active Control of Jet Noise

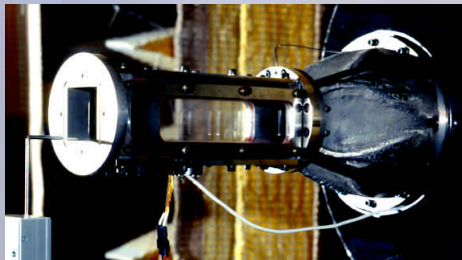
- Control of jet instability modes
- Jet shaping
- Piezoelectric and glow discharge

Fixed Geometry  
Nozzle



Fluidic Jet  
Shaping Control

Effective  
Jet Shapes



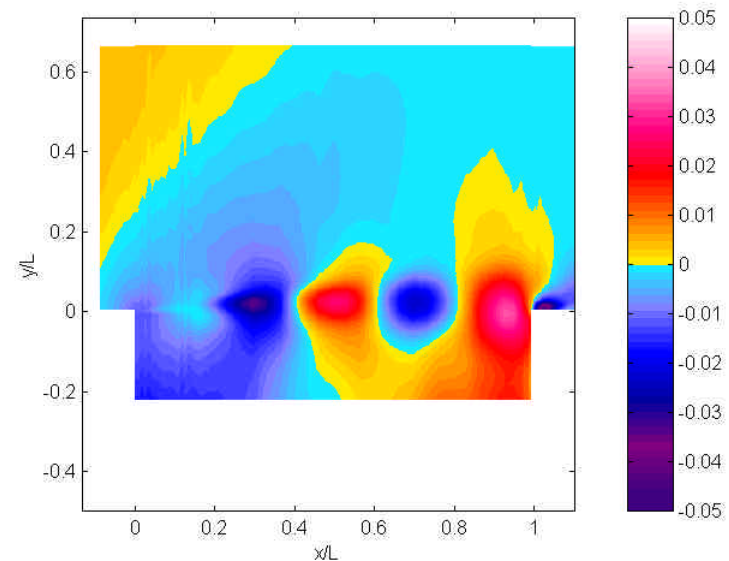
AB - Kevin Kinzie

## Closed-Loop Control of Cavity Shear Layer Instabilities

- Multiple mode control



Mode 3 in Mach 0.6 cavity



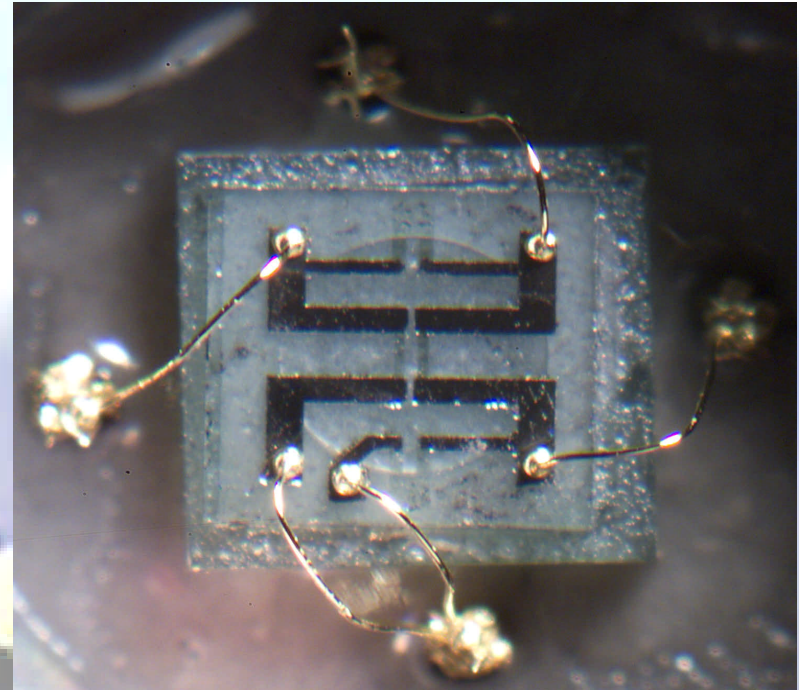
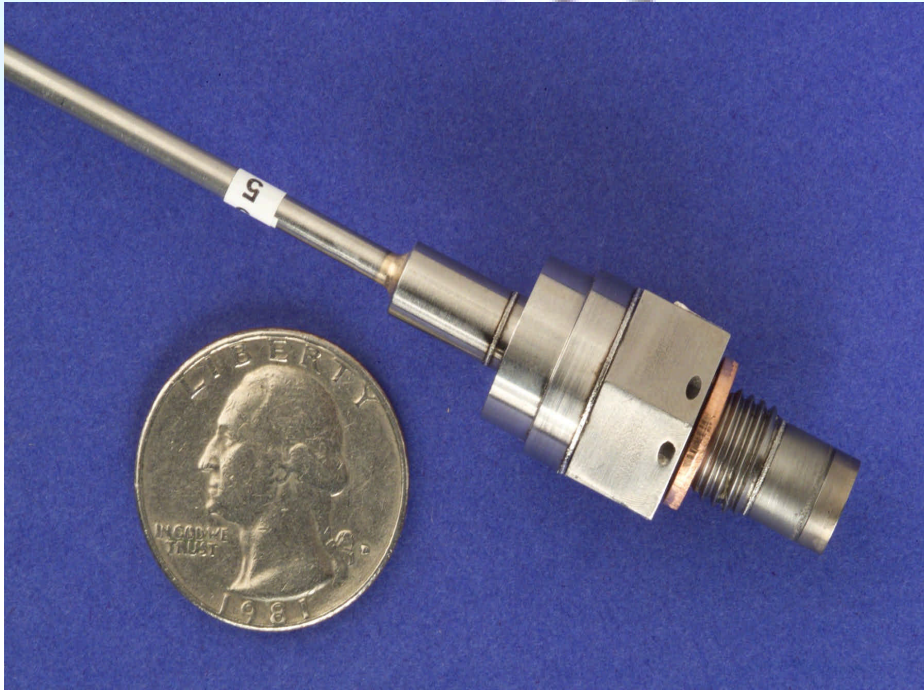
FPCB - Mike Kegerise



# Hyper-X FFS/HXFE Yaw Effects Test



# HOTPC TECHNICAL ACCOMPLISHMENT

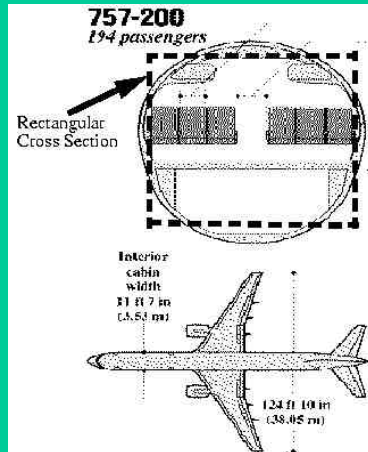


- High Temperature SiC technology developed by the HOTPC Project at GRC
- Packaging and Laboratory Demonstration performed by Kulite Semiconductor Products through an SBIR
- Engine and Test Provided by Honeywell through PIWG (Propulsion Instrumentation Working Group) that fosters pre-competitive collaboration between aircraft engine companies and the federal government on the development of advanced instrumentation.

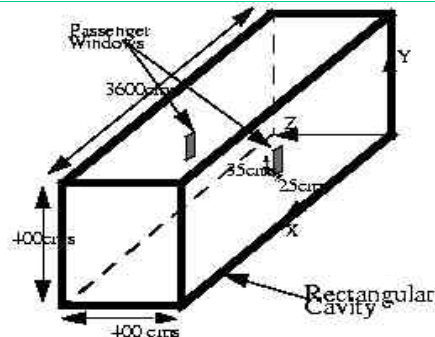


# Computational Electromagnetics

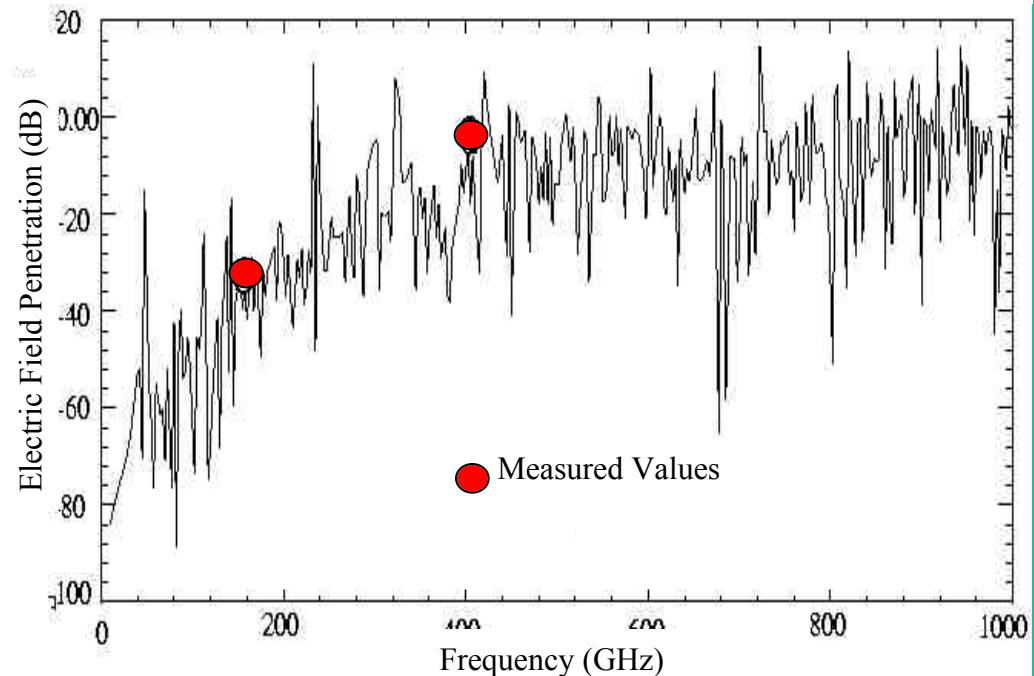
## Validation of Modal/MoM code for Field Penetration



Model for Field Penetration



Radar Field Penetration into Cabin of B757



- **What Has Been Accomplished?**
  - New technique and code development for efficient modeling of EM field penetration through holes in very large cavities. Code has been validated with measurements inside actual aircraft cabin region. Code was used in TWA-800 investigation report, which was selected for H.J.E. Reid award.
- **Significance?**
  - Capability exists for extensive aircraft safety studies related to EM field upset potential.

# Final Demonstrations of TAP Technologies

PCA Milestone: Completed demonstration of all TAP developed technologies and procedures (9/00)

## Field Demo of Aircraft Vortex Sensing System



## 757 Research System Flight Demonstration



## CTAS/FMS Flight and Full-Mission Simulation



## Field Demo of Low-Visibility Landing and Surface Ops



## Flight Test/Sim Demo of Airborne Info for Lateral Spacing





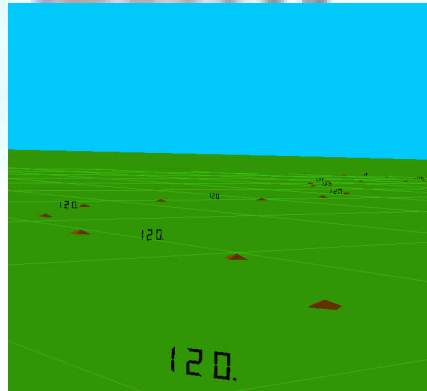
# T-NASA: A Human Factors Success Story

## R&T Base

HUD symbology  
development /  
Attention Management  
(1977-1982)



“Scene-Linked HUD  
Symbology” concept  
developed (1995)



## TAP Project

T-NASA (Taxiway  
Navigation and  
Situation Awareness)  
System (1995-1999)

### HUD



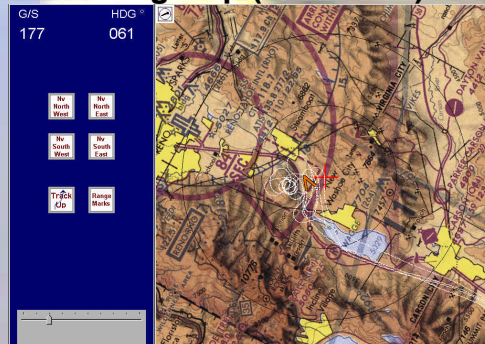
## Technology Transfer

Rockwell Collins/Flight Dynamics  
Surface Guidance System  
(HUD and Electronic Moving Map)  
Based on T-NASA concepts

In-flight Electronic  
Moving Map (1974)



Rotorcraft/Firefighter Electronic  
Moving Map (1990-1997)



ANDS - Advanced Navigational  
Display System

### Moving Map



Announced August 2000  
Phased Certification  
planned 2003/2006

# Concluding Remarks

- **Aeronautics, Launch Vehicles, and Spacecraft require Breakthrough new Technologies to meet Future Design Challenges**
- **Innovative new ideas are required to expand capacity, improve safety, reduce community noise, reduce emissions and maintain an affordable air travel system for all sectors of our society**
- **System trade studies are required to define requirements and to assess potential payoff of new technologies to guide R&D investment strategies.**
- **National Investment in High Risk Innovative New Technologies for Aerospace Applications must be Increased to Insure Successful Development of Vehicles and Systems that will Improvement the Quality of life of all Americans.**